

# **Applied Thermodynamics**

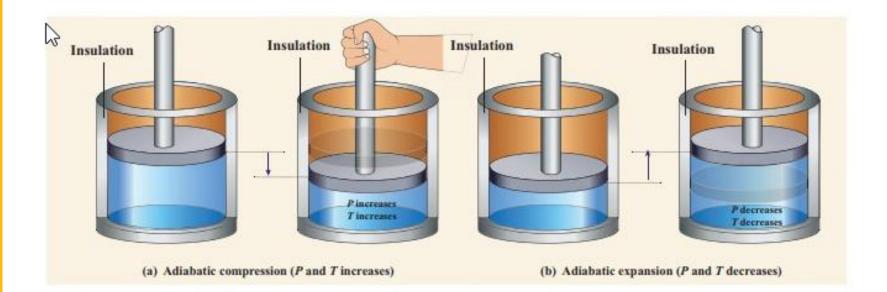
# Week\_09

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# **Mechanical Engineering Department**

## **Reversible adiabatic non-flow processes:**

An adiabatic process is one in which no heat is transferred to or from the system. Such a process can be reversible or irreversible.



REVERSIBLE ADIABATIC PROCESS FOR VAPOUR: Let us consider a reversible a diabatic non flow Process for a vapour undergoing from state point () to state point (2) as show in Fig below INSULATION .

#### **Reversible adiabatic non-flow processes:**

From non-flow energy equation me have:  $A + M = U_2 - U_1 \Rightarrow M = U_2 - U_1 \rightarrow 0$ Equation () is valid for an adiabatic non zlow process whether or not The process is reversible. For a vapour undergoing a reversible adiabatic process The work done can be found by evaluating 'l' and 'l'2 from steam Tables. NOTE: In an adiabatic compression process all the work done on the fluid goes to increasing the internal Energy of the fluid, while in an adiabatic Expansion process the work done by The fluid is at an expense of reduction in internal Energy of The fluid.

REVERSIBLE ADIABATE PROCESS FOR A PERFECT GAS: Consider a reversible adabatic process for a perfect of dergoing from state point () to state point () In Fig. below.	let us gras un- as shown
GIASS INSULATION	

**Reversible adiabatic non-flow processes:** 

Irom non flow energy equation me have; Q+W=U2-U1 => dQ+dw=du=> dQ+d[-Spdn]=du. => dQ - pdre = du => dQ = du + pdre => du + pdre =0 -2 Since we know That for a perfect gas: u= QT => du = CvdT ; pv=RT=> P= RT Hence Putting all These values in equation @ me have Crolt + RT dre=0 Dividing Through by "T" to give a form That can be integrated; CV dT + R dre=0

By integrating Both sides with refrance for 
$$T''$$
  
 $C_V \ln T + R \ln v^2 = Constant \Rightarrow C_V \ln \left[\frac{Pv}{R}\right] + R \ln v^2 = Constant = 
 $\frac{C_V \ln \left[\frac{Pv}{R}\right]}{C_V} + \frac{R \ln v^2}{C_V} = \frac{Constant}{C_V} \Rightarrow \ln \left[\frac{Pv}{R}\right] + \frac{R \ln v}{C_T} = constant$   
 $\Rightarrow \ln \left[\frac{Pv}{R}\right] + \frac{R \ln v^2}{(R/6-1)} = constant \Rightarrow \ln \left[\frac{Pv}{R}\right] + (v-1)\ln v = const-
=  $\ln \left[\frac{Pv}{R}\right] + \ln (v^{v-1}) = constant =  $\ln \left[\frac{Pv}{R}\right] = constant$   
 $\Rightarrow \ln \left[\frac{Pv}{R}\right] = constant =  $\ln \left[\frac{Pv}{R}\right] = constant$$$$$ 

#### **Reversible adiabatic non-flow processes:**

 $\Rightarrow P \mathcal{V} = R \times Constant \Rightarrow P \mathcal{V} = constant \longrightarrow 3.$  We Therefore have a simple relationship b/w "P" and " $\mathcal{V}$ " for any perfect gas under groing a reversible adiabatic process. Each perfect gas has its own value of "8". From eq.3 we have;  $P_1 \mathcal{V}_1 = P_2 \mathcal{V}_2 \Rightarrow \frac{P_1}{P_2} = \left(\frac{\mathcal{V}_2}{\mathcal{V}_1}\right) \longrightarrow (A)$ 

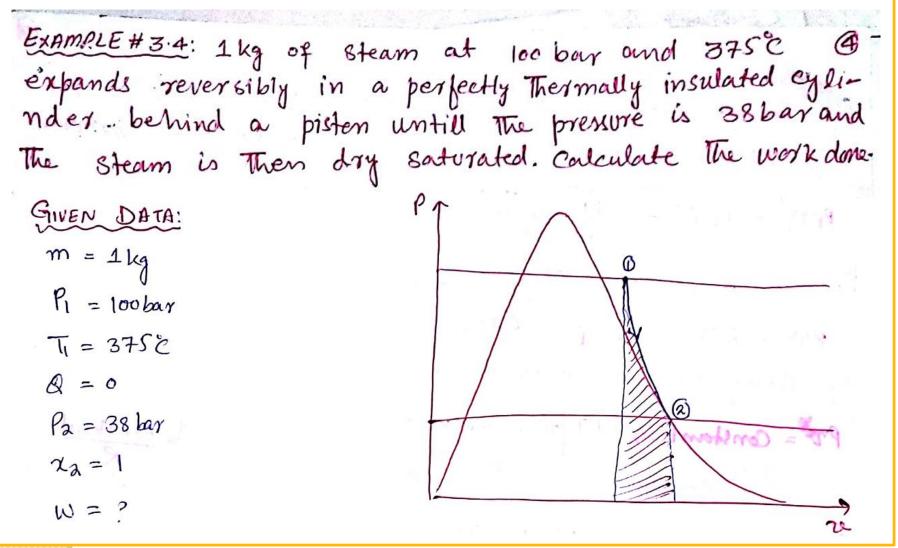
RELATIONSHIP BETINEEN "T'A 22" since we know That from eq.  

$$Pv = RT \implies P = RT/20$$
  
By Rutting This value in eq. 3 we have;  
 $Pv' = Constant \implies [RT/2]v = constant \implies Tv' = \frac{constant}{R}$   
 $\Rightarrow Tv' = Constant \implies [RT/2]v = constant \implies Tv' = \frac{constant}{R}$ 

RELATIONSHIP BETWEEN T AND 'P': Since we know That fing  

$$P2^{\circ} = RT \implies 2^{\circ} = RT/p$$
  
By Putting This value in equal we have:  
 $P2^{\circ} = Constant \implies P[RT]^{\circ} = Constant \implies T^{\circ} = Constant$   
 $\implies T^{\circ} = Constant \implies P[RT]^{\circ} = Constant \implies T^{\circ} = Constant$   
 $\implies T^{\circ} = Constant \implies P[RT]^{\circ} = Constant \implies T^{\circ} = Constant$   
 $\implies T^{\circ} = Constant \implies T^{\circ}_{P} = Constant \implies T^{\circ}_{P} = Constant$ 

From equation @ The work done for an adiabatic proces for a perfect gas can be calculated by;  $W = U_2 - U_1 = C_V (T_2 - T_1) = \binom{R}{8-1} (T_2 - T_1) = \frac{R(T_2 - T_1)}{8-1} = \frac{R(T_2$  $= \mathcal{W} = \frac{P_2 \mathcal{V}_2 - P_1 \mathcal{V}_1}{\mathcal{V}_1} \longrightarrow (\overline{T})$ The above empression is used to calculate work for a perfect gas.



**Reversible adiabatic non-flow processes:** 

# SOLUTION;

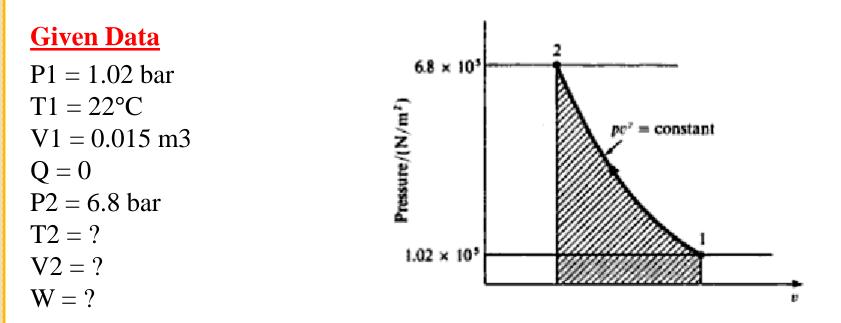
Since we know that for a reversible adiabatic process  $M = U_2 - U_1 - M$ Now from superheated steam Tables: At P\_i = 100 bar 5 T\_i = 375°C; h\_i = 3017 KJ/kg 5 2\_i = 0.02453 \frac{m^3}{100}  $\Rightarrow U = h - P2 = 3017 - (100 \times 10^5)(0.02453) = 2771.7 KJ/kg$ 

**Reversible and Irreversible Processes Reversible adiabatic non-flow processes:** Similarly: At Pa= 38bar; 2a=1; Ua=Ug= 2602 kJ/kg. Putting all These values in equation () we have; W= 2602 - 27.71.7 = -169 kJ/kg. work done by The Steam = + 169 KJ/169.

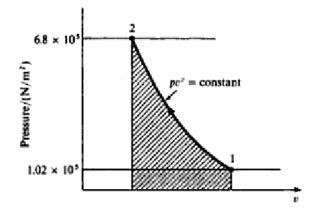
**Reversible adiabatic non-flow processes:** 

## Example 3.5

Air at 1.02 bar, 22 °C, initially occupying a cylinder volume of 0.015 m<sup>3</sup>, is compressed reversibly and adiabatically by a piston to a pressure of 6.8 bar. Calculate the final temperature, the final volume, and the work done on the mass of air in the cylinder.



$$\frac{T_1}{T_2} = \left(\frac{p_1}{p_2}\right)^{(\gamma - 1)/\gamma} \text{ or } T_2 = T_1 \times \left(\frac{p_2}{p_1}\right)^{(\gamma - 1)/\gamma}$$
  
i.e.  $T_2 = 295 \times \left(\frac{6.8}{1.02}\right)^{(1.4 - 1)/1.4}$   
 $= 295 \times 1.7195 = 507.3 \text{ K}$   
where  $T_1 = 22 + 273 = 295 \text{ K}; \gamma \text{ for air} = 1.4,$   
i.e. Final temperature = 507.3 - 273 = 234.3 °C



**Reversible adiabatic non-flow processes:** 

$$\frac{p_1}{p_2} = \left(\frac{V_2}{V_1}\right)^{\gamma}$$
 or  $\frac{V_1}{V_2} = \left(\frac{p_2}{p_1}\right)^{1/\gamma}$ 

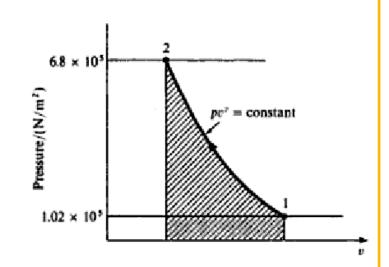
therefore

$$\frac{0.015}{V_2} = \left(\frac{6.8}{1.02}\right)^{1/1.4} = 3.877$$

therefore

$$V_2 = \frac{0.015}{3.877} = 0.003 \ 87 \ \mathrm{m}^3$$

i.e. Final volume =  $0.003 87 \text{ m}^3$ 



**Reversible adiabatic non-flow processes:** 

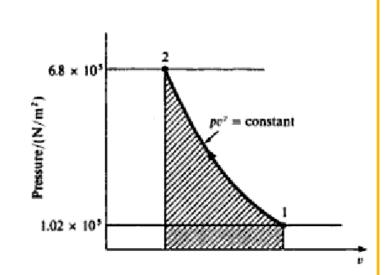
$$W = u_2 - u_1$$

$$W = c_v(T_2 - T_1) = 0.718(507.3 - 295)$$

 $= 152.4 \, kJ/kg$ 

Work input = 152.4 kJ/kg

$$m = \frac{p_1 v_1}{RT_1} = \frac{1.02 \times 10^5 \times 0.015}{0.287 \times 10^3 \times 295} = 0.0181 \text{ kg}$$
  
Total work done = 0.0181 × 152.4 = 2.76 kl



## **Polytropic Process:**

> A polytropic process is a <u>thermodynamic process</u> that obeys the relation:

$$pv^n = cons \tan t \Rightarrow p = cv^{-n}$$

The exponent n may have any value from minus infinity to plus infinity depending on the process. The above correlation can also be written as;

$$p_1 v_1^n = p_2 v_2^n \Rightarrow \frac{p_1}{p_2} = (\frac{v_2}{v_1})^n$$

The above equation is true for any working substance undergoing a reversible polytropic process.

## **Polytropic Process for Vapors:**

For any reversible non flow process the work done can be expressed as;

 $W = -\int p dv = -\int cv^{-n} dv$  $W = c \frac{v_2^{-n+1} - v_1^{-n+1}}{n-1} = \frac{p_2 v_2 v_2^{-n+1} - p_1 v_1 v_1^{-n+1}}{n-1}$  $W = \frac{p_2 v_2^n v_2^{-n+1} - p_1 v_1^n v_1^{-n+1}}{n-1} = \frac{p_2 v_2 - p_1 v_1}{n-1} \to (1)$ 

## **Polytropic Process for a Perfect Gas:**

Consider now a polytropic process for a perfect gas;

 $pv^n = cons \tan t \Rightarrow p = cv^{-n}$ 

Since 
$$pv = RT \Rightarrow p = \frac{RT}{v}$$

$$\Rightarrow \frac{RT}{v}v^{n} = constant \Rightarrow \frac{T}{v}v^{n} = \frac{constant}{R} \Rightarrow Tv^{n-1} = contsnat$$

$$\Rightarrow T_1 v_1^{n-1} = T_2 v_2^{n-1} \Rightarrow \frac{T_1}{T_2} = (\frac{v_2}{v_1})^n$$

## **Polytropic Process for a Perfect Gas:**

Relation between Temperature (T) and Pressure (p);

$$pv^n = cons \tan t \Rightarrow p = cv^{-n}$$

Since 
$$pv = RT \Rightarrow v = \frac{RT}{p}$$

$$\Rightarrow p\left[\frac{RT}{p}\right]^{n} = constant \Rightarrow \frac{T}{p^{\frac{n-1}{n}}} = \frac{constant}{R^{-n}} \Rightarrow \frac{T}{p^{\frac{n-1}{n}}} = contsnat$$

$$\Rightarrow \frac{T_1}{p_1^{\frac{n-1}{n}}} = \frac{T_2}{p_2^{\frac{n-1}{n}}} \Rightarrow \frac{T_1}{T_2} = \frac{p_1^{\frac{n-1}{n}}}{p_2^{\frac{n-1}{n}}} \Rightarrow \frac{T_1}{T_2} = \left[\frac{p_1}{p_2}\right]^{\frac{n-1}{n}}$$

### **Polytropic Process for a Perfect Gas:**

➢ For a perfect gas expanding polytropically it is sometimes more convenient to express the work input in terms of the temperatures at the end states.

$$W = \frac{p_2 v_2 - p_1 v_1}{n - 1} = \frac{RT_2 - RT_1}{n - 1} = \frac{R(T_2 - T_1)}{n - 1}$$

$$W = \frac{P_2 V_2 - P_1 V_1}{n - 1} = \frac{mRT_2 - mRT_1}{n - 1} = \frac{mR(T_2 - T_1)}{n - 1}$$

 $\succ$  Using the non-flow energy equation, the heat flow during the process can be found:

$$Q + W = u_2 - u_1 = c_v (T_2 - T_1)$$

$$Q + \frac{R(T_2 - T_1)}{n - 1} = c_v (T_2 - T_1)$$
$$Q + \frac{R(T_2 - T_1)}{n - 1} = \left[\frac{R}{(\gamma - 1)}\right] (T_2 - T_1)$$

#### **Polytropic Process for a Perfect Gas:**

$$Q + \frac{R(T_2 - T_1)}{n - 1} = \left[\frac{R}{(\gamma - 1)}\right](T_2 - T_1)$$

$$Q = \left[\frac{R}{(\gamma - 1)}\right](T_2 - T_1) - \frac{R(T_2 - T_1)}{n - 1}$$

$$Q = R(T_2 - T_1) \left[ \frac{1}{(\gamma - 1)} - \frac{1}{n - 1} \right]$$

$$Q = R(T_2 - T_1) \left[ \frac{n - 1 - \gamma + 1}{(\gamma - 1)(n - 1)} \right]$$

$$Q = R(T_2 - T_1) \left[ \frac{n - \gamma}{(\gamma - 1)(n - 1)} \right]$$

$$Q = \left[\frac{n-\gamma}{(\gamma-1)}\right] \frac{R(T_2 - T_1)}{(n-1)} = \left[\frac{n-\gamma}{(\gamma-1)}\right] W$$

The above equation is a convenient and concise expression relating the heat supplied and the work input in a polytropic process.

- In a compression process work is done on the gas, and hence the term W is positive. Therefore it can be seen from the equation that when the polytropic index n is greater than y, in a compression process, then the right-hand side of the equation is positive (i.e. heat is supplied during the process). Conversely, when n is less than y in a compression process, then heat is rejected by the gas.
- Similarly, the work input in an expansion process is negative, therefore when n is greater than γ, in an expansion process, heat is rejected; and when n is less than γ, in an expansion process, heat must be supplied to the gas during the process. for all perfect gases has a value greater than unity.

## **Polytropic Process for a Perfect Gas:**

#### Example 3.6:

At the commencement of compression in the reciprocating compressor of a refrigeration plant the refrigerant is dry saturated at 1 bar. The compression process follows the law  $pv^{1.1}$  = constant until the pressure is 10 bar. Using the properties of refrigerant given in Table interpolating where necessary, calculate:

(i) the work done on the refrigerant during the process;

(ii) the heat transferred to or from the cylinder walls during the process.

Saturation values				Superheat values at 10 bar		
P.	t.,	v,	hr	h <sub>s</sub>	v	h
(bar)	(°C)	(m <sup>3</sup> /kg)	(kJ/kg)		(m <sup>3</sup> /kg)	(kJ/kg)
1 10	- 30 42	0.160 0.018	8.9 76.3	174.2 203.8	0.020	224.0

**Polytropic Process for a Perfect Gas:** 

Given Data:

P1=1 bar X1=1  $Pv^{1.1} = constant$  P2=10 bar W=? Q=?

Solution (i) From Table 3.1,  $v_1 = v_{g1} = 0.16 \text{ m}^3/\text{kg}$ . We then have

$$\Rightarrow \frac{p_1}{p_2} = \left(\frac{\nu_2}{\nu_1}\right)^n \Rightarrow \qquad \frac{\nu_2}{\nu_1} = \left(\frac{p_1}{p_2}\right)^{1/1.1} = \left(\frac{1}{10}\right)^{0.909} = 0.1233$$

therefore

i.c.

$$v_2 = 0.1233 \times 0.16 = 0.01973 \text{ m}^3/\text{kg}$$

From equation (3.24)

$$W = \frac{p_2 v_2 - p_1 v_1}{n - 1}$$
  
=  $\left\{ \frac{(10 \times 0.01973) - (1 \times 0.16)}{1.1 - 1} \right\} \times 10^5$   
= 37 300 N m = 37.3 kJ  
Work done on the refrigerant = 37.3 kJ

### **Polytropic Process for a Perfect Gas:**

(ii) To find the heat transferred it is first necessary to evaluate the internal energies at the end states. Using equation (1.9), h = u + pv, we have,

$$u_{1} = u_{g1} = h_{g1} - p_{g1}v_{g1}$$
$$= 174.2 - \left(\frac{1 \times 10^{5} \times 0.16}{10^{3}}\right) = 158.2 \text{ kJ/kg}$$

Interpolating from superheat tables at 10 bar,  $v_2 = 0.01973 \text{ m}^3/\text{kg}$  we have

$$h_2 = 203.8 + \frac{(0.01973 - 0.018)}{(0.02 - 0.018)} \times (224 - 203.8)$$
$$= 221.3 \text{ kJ/kg}$$

**Polytropic Process for a Perfect Gas:** 

Then using equation (1.9)

 $u_2 = 221.3 - \left(\frac{10 \times 10^5 \times 0.01973}{10^3}\right) = 201.6 \text{ kJ/kg}$ 

From equation (1.4)

$$Q+W=(u_2-u_1)$$

therefore

$$Q = -37.3 + (201.6 - 158.2) = 6.1 \text{ kJ/kg}$$

i.e. the heat transferred from the cylinder walls to the refrigerant during the compression process is 6.1 kJ/kg.

**Polytropic Process for a Perfect Gas:** 

Example 3.7:

1 kg of a perfect gas is compressed from 1.1 bar, 27 °C according to a law  $pv^{1.3}$  = constant, until the pressure is 6.6 bar. Calculate the heat flow to or from the cylinder walls:

- (i) When the gas is ethane (molar mass 30 kg/kmol), which has  $c_p = 2.10 \text{ kJ/kg K}$ .
- (ii) When the gas is argon (molar mass 40 kg/kmol), which has  $c_p = 0.520 \text{ kJ/kg K}$ .

Given Data:

m=	1  kg $P1 = 1.1  bar$	$T1 = 27^{\circ}C$	$Pv^{1.3} = constant$
P2 =	= 6.6 bar	Q?	
i)	Gas = Ethane	$\widetilde{m}$ =30 kg/kmol	$c_p = 2.10 \frac{kJ}{kgK}$
ii)	Gas = Argon	$\widetilde{m}$ =40 kg/kmol	$c_p = 0.520 \ kJ/kgK$

### **Polytropic Process for a Perfect Gas:**

Solution From equation (3.29), for both ethane and argon,

$$\frac{T_1}{T_2} = \left(\frac{p_1}{p_2}\right)^{(n-1)/n} \text{ or } T_2 = T_1 \left(\frac{p_2}{p_1}\right)^{(n-1)/n}$$
  
i.e. 
$$T_2 = 300 \left(\frac{6.6}{1.1}\right)^{1.3 - 1/1.3} = 300 \times 6^{0.231} = 300 \times 1.512 = 453.6 \text{ K}$$

where  $T_1 = 27 + 273 = 300$  K.

### **Polytropic Process for a Perfect Gas:**

(i) From equation (2.9),  $R = \tilde{R}/\tilde{m}$ , therefore, for ethane

$$R = \frac{8.3145}{30} = 0.277 \text{ kJ/kg K}$$

Then from equation (2.17),  $c_p - c_v = R$ , therefore

$$c_v = 2.10 - 0.277 = 1.823 \text{ kJ/kg K}$$

where  $c_p = 1.75 \text{ kJ/kg K}$  for ethane. Then from equation (2.20)

$$\gamma = \frac{c_p}{c_v} = \frac{2.10}{1.823} = 1.152$$

From equation (3.30)

$$W = \frac{R(T_2 - T_1)}{n - 1} = \frac{0.277 \times (453.6 - 300)}{1.3 - 1} = 141.8 \text{ kJ/kg}$$

Then from equation (3.32)

$$Q = \left(\frac{n-\gamma}{\gamma-1}\right)W = \left(\frac{1.3 - 1.152}{1.152 - 1}\right) \times 141.8 = 138.1 \text{ kJ/kg}$$

i.e. Heat supplied = 138.1 kJ/kg

#### **Polytropic Process for a Perfect Gas:**

(ii) Using the same method for argon we have

$$R = \frac{8.3145}{40} = 0.208 \text{ kJ/kg K}$$

Also  $c_v = 0.520 - 0.208 = 0.312 \text{ kJ/kg K}$ 

therefore

$$\gamma = \frac{0.520}{0.312} = 1.667$$

Then the work input is given by

$$W = \frac{R(T_1 - T_2)}{n - 1} = \frac{0.208 \times (453.6 - 300)}{1.3 - 1} = 106.5 \text{ kJ/kg}$$
  
Then,  $Q = \left(\frac{n - \gamma}{\gamma - 1}\right)W = \left(\frac{1.3 - 1.667}{1.667 - 1}\right) \times 106.5 = -58.6 \text{ kJ/kg}$   
i.e. Heat rejected = 58.6 kJ/kg

#### **Reversible Flow Processes:**

Although flow processes in practice are usually highly irreversible, it is sometimes convenient to assume that a flow process is reversible in order to provide an ideal comparison. An observer travelling with the flowing fluid would appear to see a change in thermodynamic properties as in a non-flow process. For example, in a reversible adiabatic process for a perfect gas, an observer travelling with the gas would appear to see a process  $pv^{\gamma} = \text{constant}$  taking place, but the work input would not be given by  $-\int p \, dv$ , or by the change in internal energy as given by equation (3.13). Some work is done by virtue of the forces acting between the moving gas and its surroundings. For example, for a reversible adiabatic flow process for a perfect gas, from the flow equation (1.10), for unit mass flow rate

$$\left(h_1 + \frac{C_1^2}{2}\right) + \mathcal{Q} + W = \left(h_2 + \frac{C_2^2}{2}\right)$$

Then since 
$$Q = 0$$

$$W = (h_2 - h_1) + \left(\frac{C_2^2}{2} - \frac{C_1^2}{2}\right)$$

Also, since the process is assumed to be reversible, then for a perfect gas,  $pv^{\gamma} = \text{constant}$ . This equation can be used to fix the end states. Note that even if the kinetic energy terms are negligibly small the work input in a reversible adiabatic flow process between two states is not equal to the work input in a reversible adiabatic non-flow process between the same states (given by equation (3.13) as  $W = u_2 - u_1$ ).

#### **Reversible Flow Processes:**

Although flow processes in practice are usually highly irreversible, it is sometimes convenient to assume that a flow process is reversible in order to provide an ideal comparison. An observer travelling with the flowing fluid would appear to see a change in thermodynamic properties as in a non-flow process. For example, in a reversible adiabatic process for a perfect gas, an observer travelling with the gas would appear to see a process  $pv^{\gamma} = \text{constant}$  taking place, but the work input would not be given by  $-\int p \, dv$ , or by the change in internal energy as given by equation (3.13). Some work is done by virtue of the forces acting between the moving gas and its surroundings. For example, for a reversible adiabatic flow process for a perfect gas, from the flow equation (1.10), for unit mass flow rate

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Then since 
$$Q = 0$$

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### **Reversible Flow Processes :**

Example 3.8:

A gas turbine receives gases from the combustion chamber at 7 bar and 650 °C, with a velocity of 9 m/s. The gases leave the turbine at 1 bar with a velocity of 45 m/s. Assuming that the expansion is adiabatic and reversible in the ideal case, calculate the power output per unit mass flow rate. For the gases take  $\gamma = 1.333$  and  $c_p = 1.11$  kJ/kg K.

Given Data:

P1 = 7.0  bar	$T1 = 650^{\circ}C$	C1 = 9 m/s	P2 = 1.0  bar	C2 = 45  m/s
<b>W</b> =?	$\gamma = 1.333$	$c_p =$	$1.11 \frac{kJ}{kgK}$	

#### **Reversible Flow Processes :**

Solution Using the flow equation for an adiabatic process

$$W = \dot{m} \left\{ (h_2 - h_1) + \left( \frac{C_2^2 - C_1^2}{2} \right) \right\}$$

For a perfect gas from equation (2.18),  $h = c_p T$ , therefore,

$$W = \dot{m} \left\{ c_p (T_2 - T_1) + \left( \frac{C_2^2 - C_2^2}{2} \right) \right\}$$

To find  $T_2$  use equation (3.21),

$$\frac{T_1}{T_2} = \left(\frac{p_1}{p_2}\right)^{(\gamma-1)/\gamma}$$
  
i.e. 
$$\frac{T_1}{T_2} = \left(\frac{7}{1}\right)^{(1.333-1)/1.333} = 1.626$$

therefore

$$T_2 = \frac{T_1}{1.626} = \frac{923}{1.626} = 567.7 \text{ K}$$

where  $T_1 = 650 + 273 = 923$  K.

### **Reversible Flow Processes :**

Hence substituting for unit mass flow rate

$$\dot{W} = 1 \times 1.11(567.7 - 923) + \left(\frac{45^2 - 9^2}{2 \times 10^3}\right)$$

therefore

$$\dot{W} = -394.4 + 0.97 = -393.4 \,\mathrm{kW}$$

**Exercise Problems:** 3.8, 3.9, 3.10, 3.11, 3.12, 3.13, 3.14, 3.15

